

Harvard-Smithsonian Center for Astrophysics

Precision Astronomy Group

To: Distribution
 From: J.D. Phillips
 Subject: Transverse tilts of the FAME compound mirror

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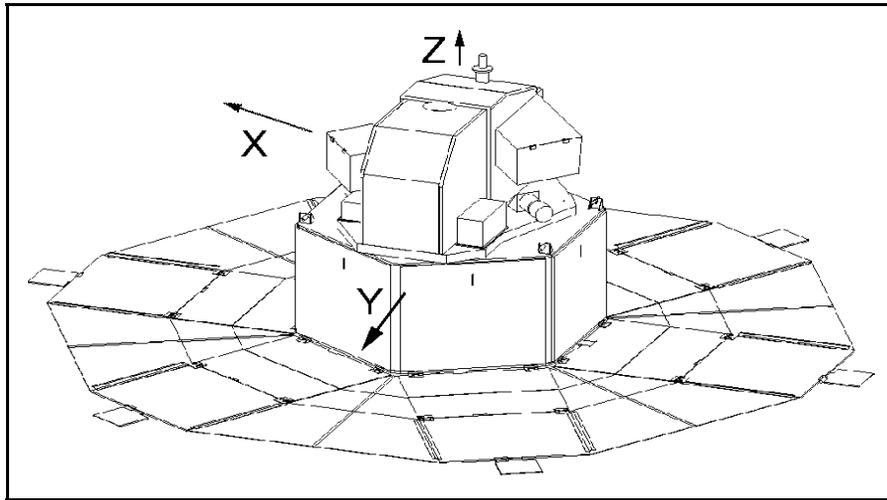


Figure 1. Instrument coordinates [FAME Concept Study Report, p. 4-72].

The FAME basic angle γ (nominally 81.5°) is established by the compound mirror, a pair of plane mirrors mounted very stably at a relative angle of $\gamma/2$. Their normals lie nominally in the instrument xy plane (see Fig. 1). This memo addresses the effect of a relative "out of plane" tilt (defined below). Rotation of the compound mirror as a whole affects the alignment of

the focal plane to the rotation axis, and will need to be treated elsewhere.

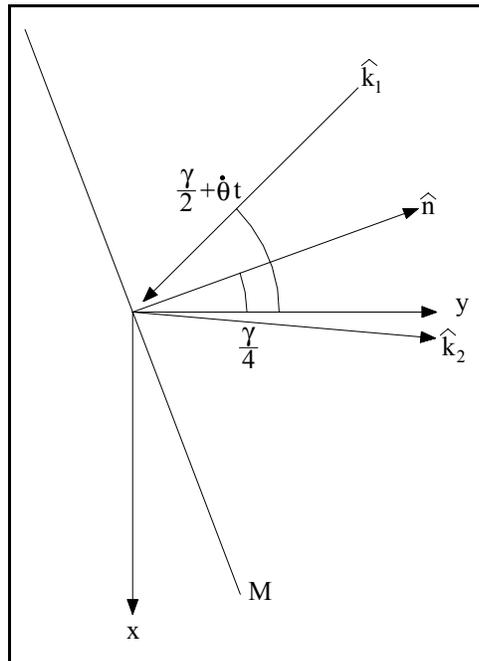


Figure 2. Rays at compound mirror. M is the surface of one element of the compound mirror

The tilt considered here is a difference in the z-components of the normals of the two elements of the compound mirror, called here a relative "out of plane" tilt. The primary effect of such a tilt is to make the images of stars entering the two ports travel at different angles across the focal plane. A secondary effect is to make the images travel at different speeds on the focal plane. For reasonable tilts, the latter is insignificant.

Consider a beam entering the port to the right in Fig. 1 in the xy plane, with propagation vector \hat{k}_1 (which has unit length). It reflects at the corresponding element of the compound mirror, whereafter it has propagation vector \hat{k}_2 (Fig. 2). Take the mirror normal to be tilted by an angle ϵ . It is

$$\hat{\mathbf{n}} = \frac{1}{\sqrt{1 + \varepsilon^2}} \begin{pmatrix} -\sin \frac{\gamma}{4} \\ \cos \frac{\gamma}{4} \\ \varepsilon \end{pmatrix} . \quad (1)$$

The input propagation vector is

$$\hat{\mathbf{k}}_1 = \begin{pmatrix} \sin(\frac{\gamma}{2} + \dot{\theta}t) \\ -\cos(\frac{\gamma}{2} + \dot{\theta}t) \\ 0 \end{pmatrix} , \quad (2)$$

where $\dot{\theta}$ is the instrument rotation rate, and $t=0$ when the star is in the center of the field of view. The reflected ray $\hat{\mathbf{k}}_2$ is given by

$$\hat{\mathbf{k}}_2 = \hat{\mathbf{k}}_1 - 2(\hat{\mathbf{k}}_1 \cdot \hat{\mathbf{n}}) \hat{\mathbf{n}} . \quad (3)$$

With the aid of Maple, using the Matlab Extended Symbolic Toolbox interface, this becomes¹

$$\hat{\mathbf{k}}_2 = \frac{1}{1 + \varepsilon^2} \begin{pmatrix} \sin(\dot{\theta}t) + \varepsilon^2 \sin(\frac{\gamma}{2} + \dot{\theta}t) \\ \cos(\dot{\theta}t) - \varepsilon^2 \cos(\frac{\gamma}{2} + \dot{\theta}t) \\ 2\varepsilon \cos(\frac{\gamma}{4} + \dot{\theta}t) \end{pmatrix} \quad (4)$$

Taking the time derivative,

$$\frac{d\hat{\mathbf{k}}_2}{dt} = \frac{\dot{\theta}}{1 + \varepsilon^2} \begin{pmatrix} \cos(\dot{\theta}t) + \varepsilon^2 \cos(\frac{\gamma}{2} + \dot{\theta}t) \\ -\sin(\dot{\theta}t) + \varepsilon^2 \sin(\frac{\gamma}{2} + \dot{\theta}t) \\ -2\varepsilon \sin(\frac{\gamma}{4} + \dot{\theta}t) \end{pmatrix} \quad (5)$$

(As a check on the computer algebra, $|\hat{\mathbf{k}}_2| = 1$, and $|d\hat{\mathbf{k}}_2/dt| = \dot{\theta}$.) The angle of trajectories on the focal plane is

¹ This information is for technical communication only and does not constitute an endorsement of these products.

$$\varphi = \frac{\left(\frac{d\hat{k}_2}{dt}\right)_z}{\left(\frac{d\hat{k}_2}{dt}\right)_x} = -2\varepsilon \frac{\sin\left(\frac{\gamma}{4} + \dot{\theta}t\right)}{\cos(\dot{\theta}t) + \varepsilon^2 \cos\left(\frac{\gamma}{2} + \dot{\theta}t\right)} \quad (6)$$

A relative tilt, ε , of 10 arcsec, which has been Lockheed's nominal specification [Huff, 2000], results in a relative angle of the trajectories of images of stars through the two ports of 0.14 pixel per 4096, which will result in a negligible contribution to the alignment budget.

The velocity of the images is modified by a factor $1 + \varepsilon^2 \cos(\gamma/2)$ (at the field center), resulting in a position difference, after TDI of 4096 pixels, of 7×10^{-6} pixel = 1.5 μ arcsec. This is equivalent to a (static) change of the basic angle, and is utterly negligible.

Reference

Huff 2000 E-mail from L. Huff, Lockheed-Martin Corp., 13 Nov. 2000.

Distribution

FAME project, *via* website.